

Best in class CFD meshing for the aerospace sector

Vangelis Skaperdas

physics on screen

ANSA evolution

Born in the automotive industry back in the early 90s, for FEA meshing



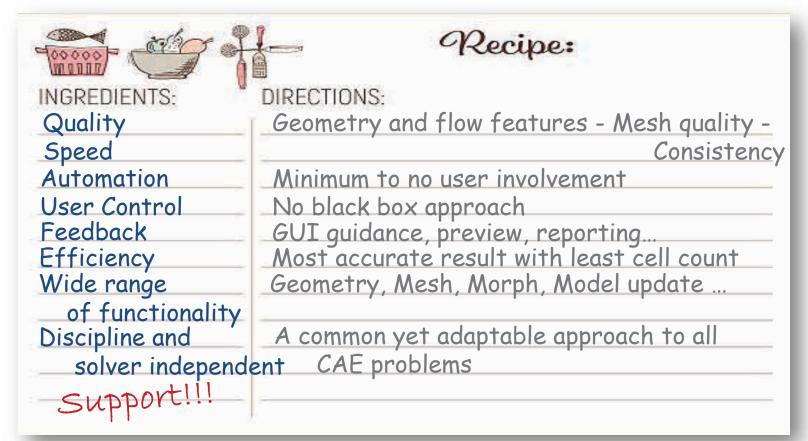
In the late 90s it becomes widespread for CFD meshing in motorsports teams and OEMs

In the late 2000s it enters the CFD aerospace sector and is growing ever since



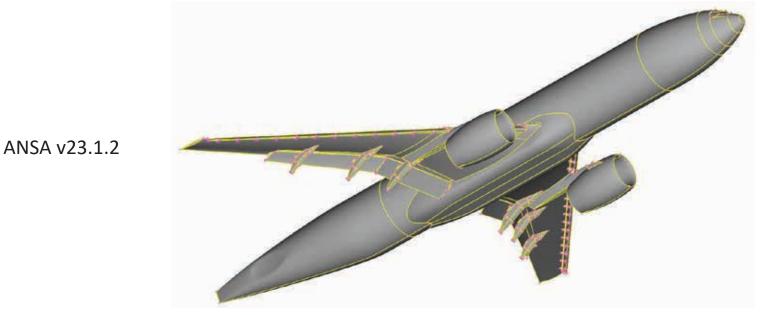


What is the recipe of success of a pre-processing software?



Case study description

 Demonstrate ANSA pre-processing capabilities on the geometry of the CRM model from the 4th High Lift Prediction Workshop



- Linux workstation with AMD Ryzen Threadripper 3970X with 32 cores (64 threads) and 256Gb RAM

ANSA steps to high fidelity mesh generation

- 1- Start ANSA in CFD layout and select your meshing profile
- 2- Read the CAD and check its topology
- 4- Assign correct orientation
- 3- Create the domain
- 5- Assign PIDs and BC types
- 6- Detect and manage Features
- 7- Setup Batch Mesh
- 8- Setup Size Field
- 9- Execute Batch Mesh
- 10- Check Mesh quality
- 11 Output mesh

Launching ANSA for CFD

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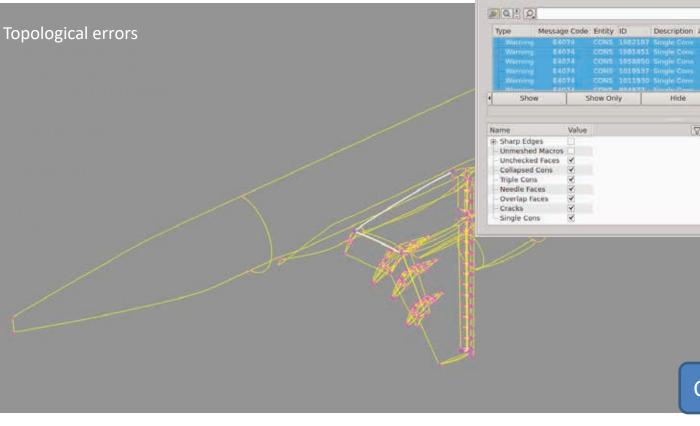
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Geometry import and check



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Check>Geometry

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Geometry import and check

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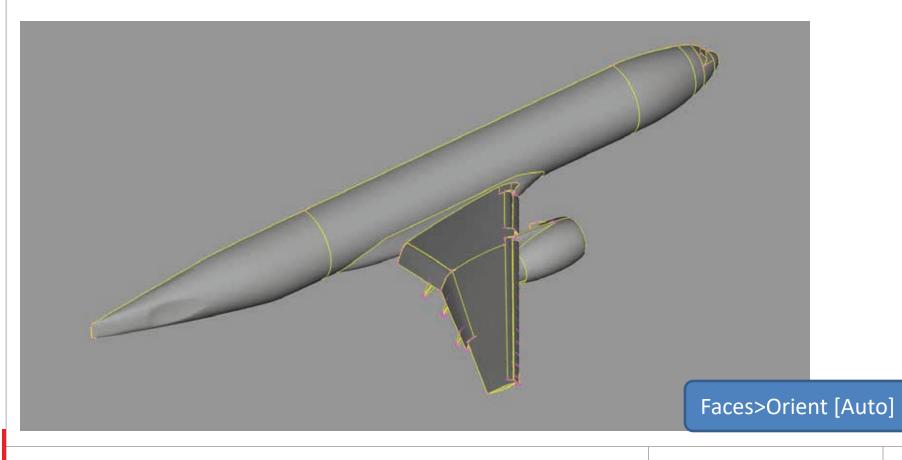
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Check>Geometry

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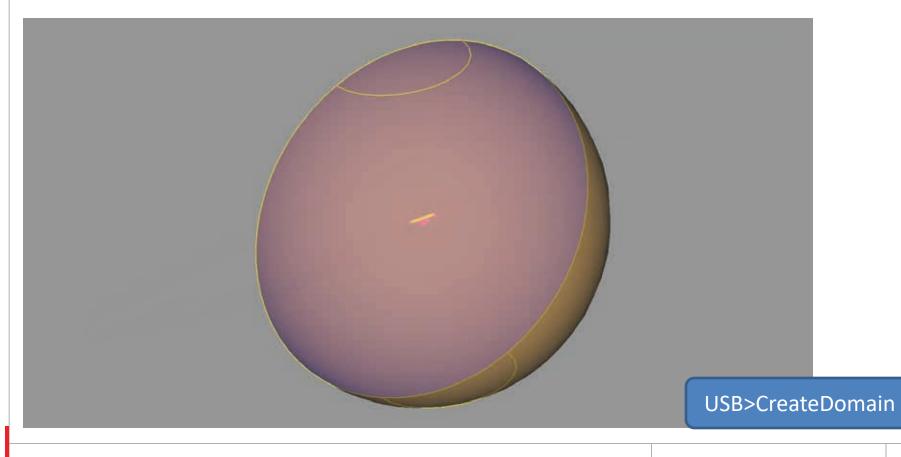
Correct orientation



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Domain creation

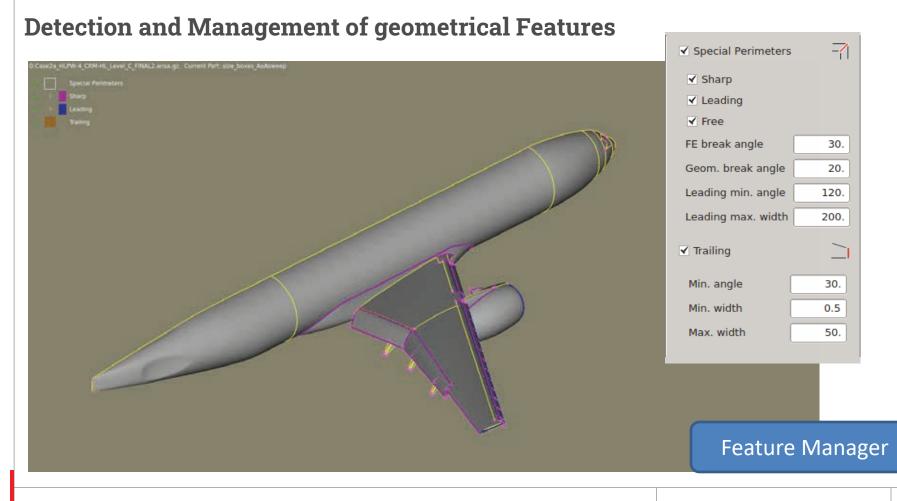


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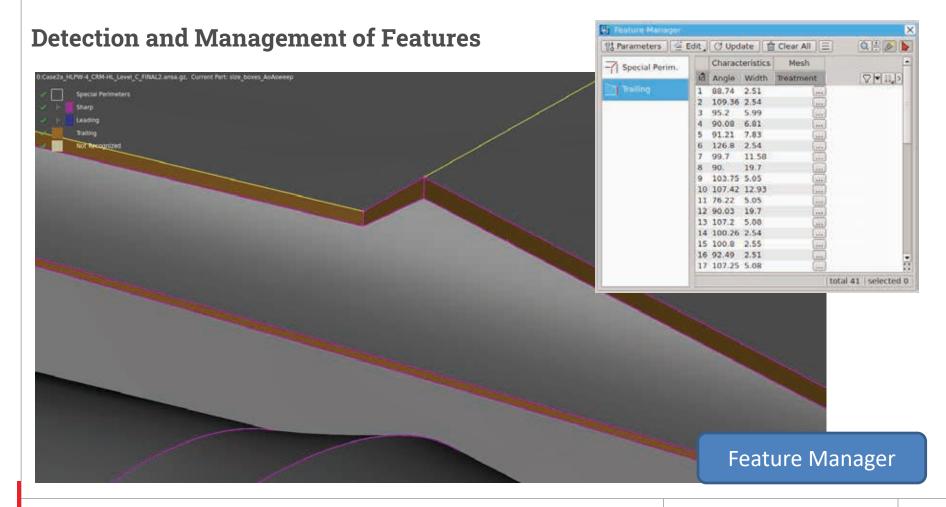


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Batch Mesh tool

Batch Mesh tool in ANSA performs fully automatic surface and volume meshing, ensuring:

- Automation
- Reusability of scenario templates for new geometries
- No user interaction and hence consistency in meshing
- Facilitates mesh refinement studies
- Mesh specs traceability as template is saved in ANSA file

New Read Scenario Autol	oad Run	Draft	result	Update status	after run	\$
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- 🖌 engine	2		CFD parameters 1	Aerospace_y+1_TAU		
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- V engine	3		CFD parameters			
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Setup of Batch Mesh tool

Case2a_HLPW-4_CRM-HL_Level_C_FINAL2 ansa.gz, Current Part: size_boxes_AoAoweep

Batch Session colors

New , Read Scenario Auto	oad Run	Draft	result	Update status	after run	•
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- ✓ lower_wing	1		CFD parameters 1	Aerospace y+1 TAU		
✓ Fuselage	1		CFD parameters 5	Aerospace y+1_TAU		
- 😪 Farfield	1		CFD parameters 4	Aerospace y+1_TAU		
✓ Symmetry	1		CFD parameters 2			
Default Session	0		CFD parameters	Aerospace y+1 TAU		
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B Default Session	18	100	CFD parameters	Aerospace_y+1_TAU		
✓ fuselage	1		CFD parameters			
1 wing	14		CFD parameters			
✓ engine	3		CFD parameters			
Volume Scenario 3	0	-				
✓ Default Session	0		CFD parameters	Aerospace y+1 TAU		

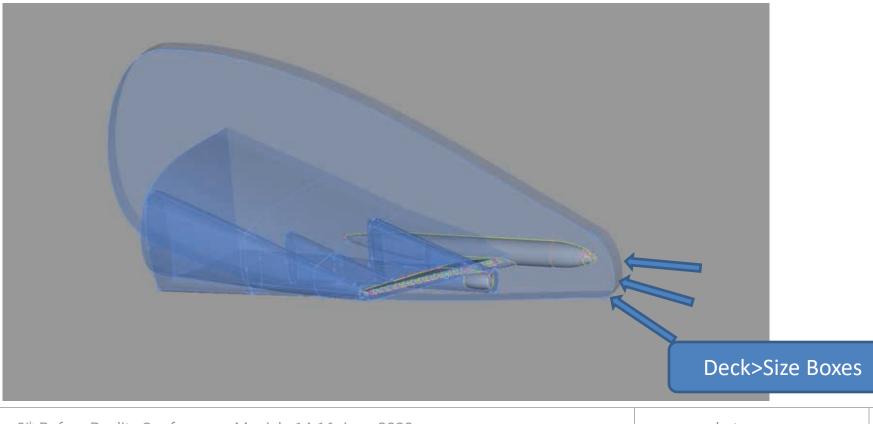
- Curvature refinement
- Sharp edge feature refinement
- Leading and trailing edges anisotropic meshing
- Proximity refinement

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Setup of additional size refinement zones

Creation of flexible Size Boxes to control the wake refinement

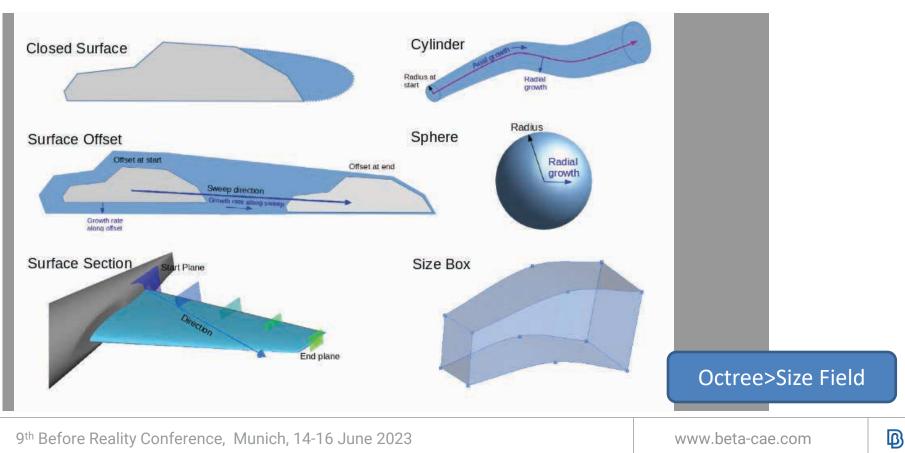


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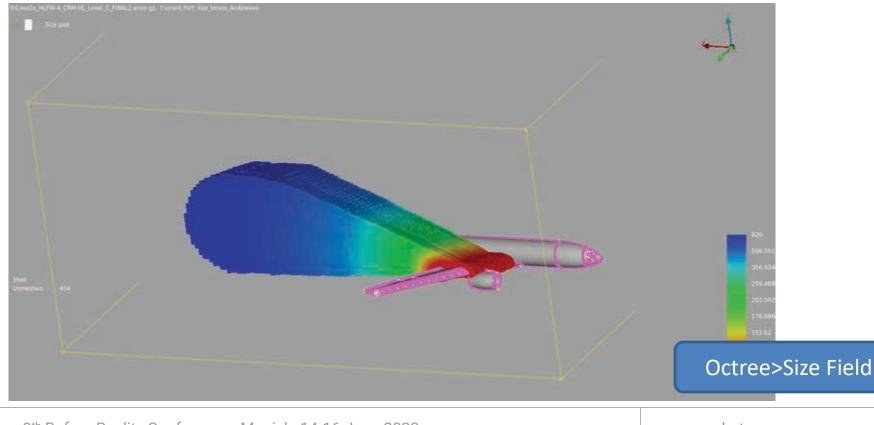
Setup of additional size refinement zones

Specification of advanced Size Field functions



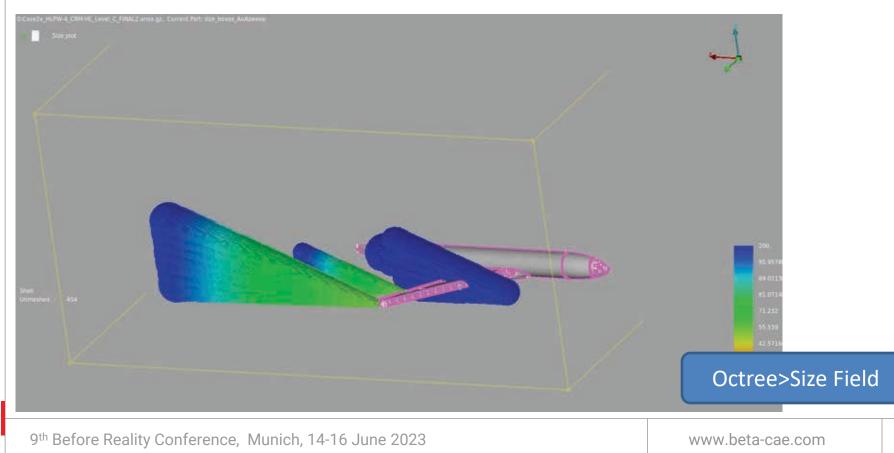
Setup of Size Field

Surface Offset Rule from selected PIDs along a user specified direction

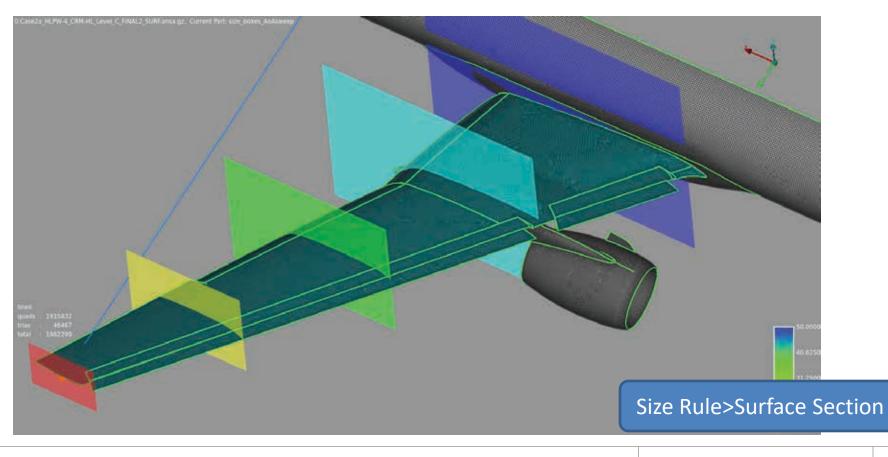


Setup of Size Field

Cylinder Rule along selected 3D Curves



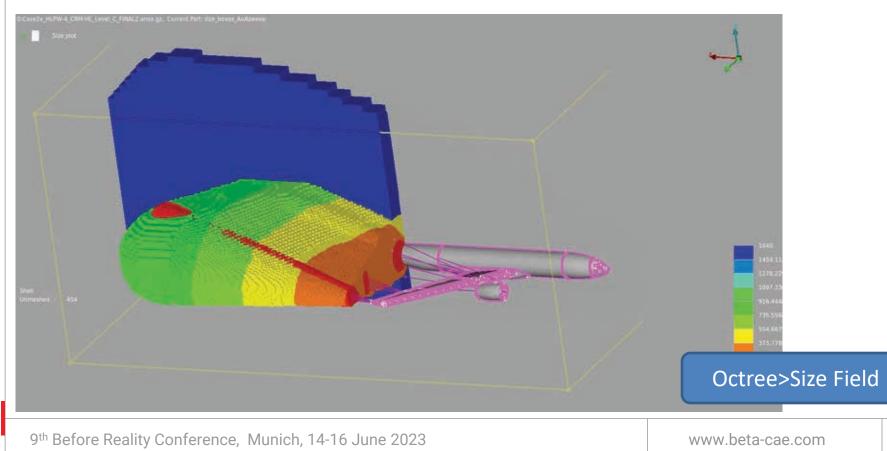
Main wing – Linear Variation of maximum length



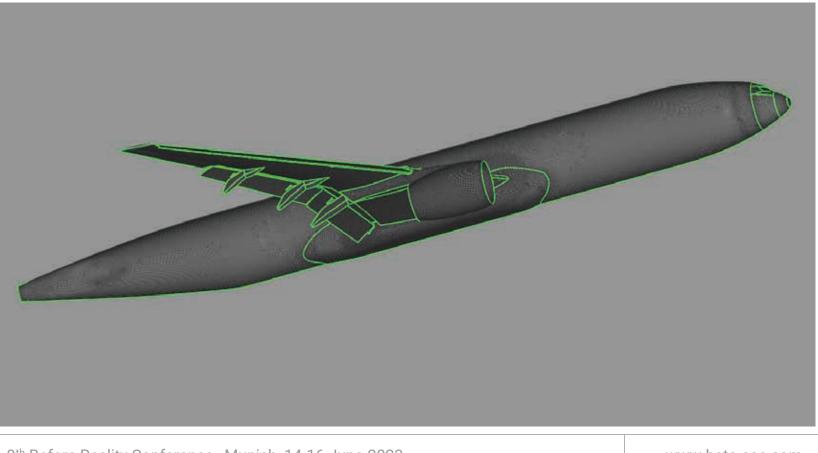
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Setup of Size Field

Final combined Size Field



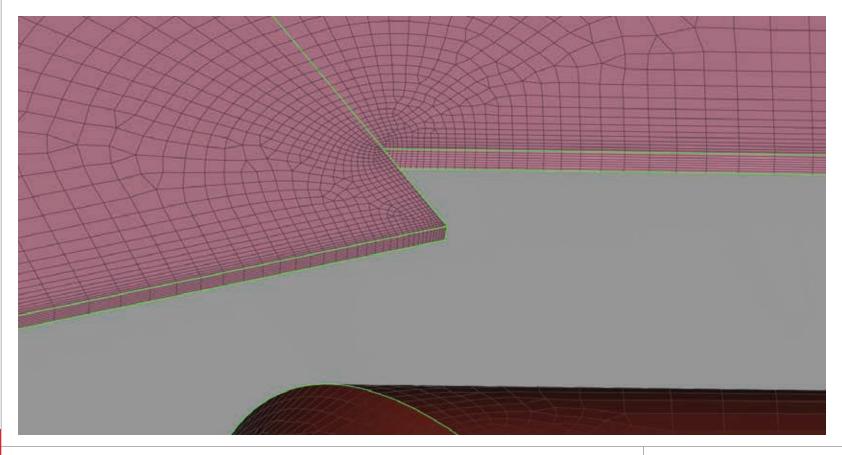
Surface mesh Level C refinement - 2 million shells



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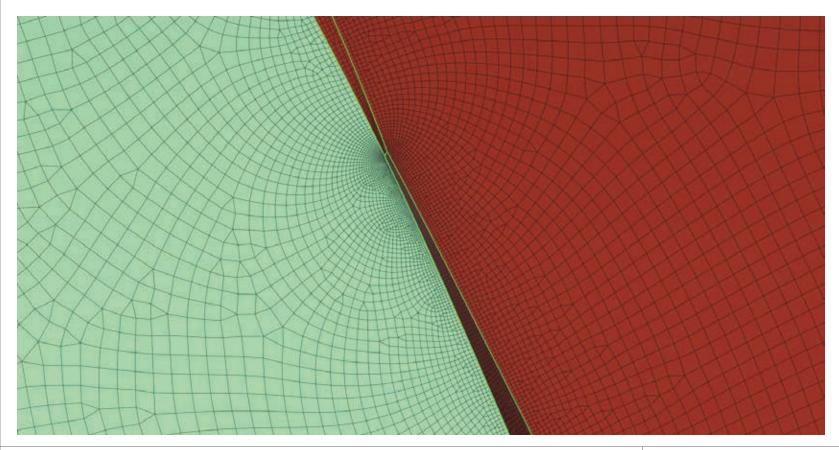
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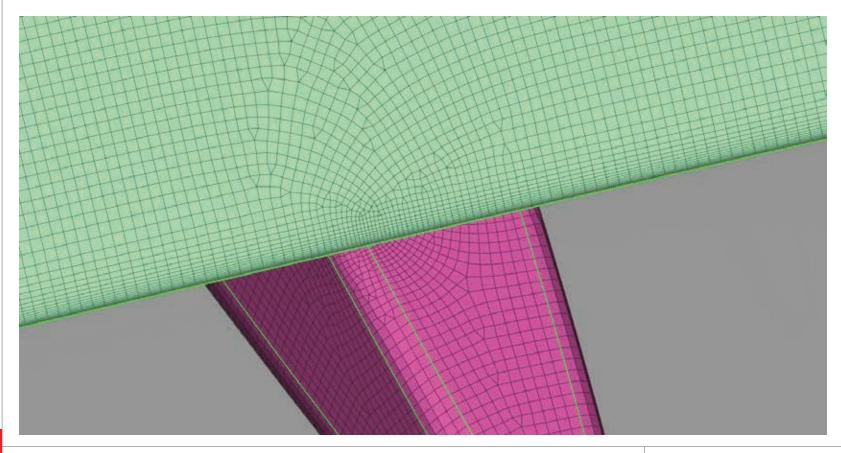


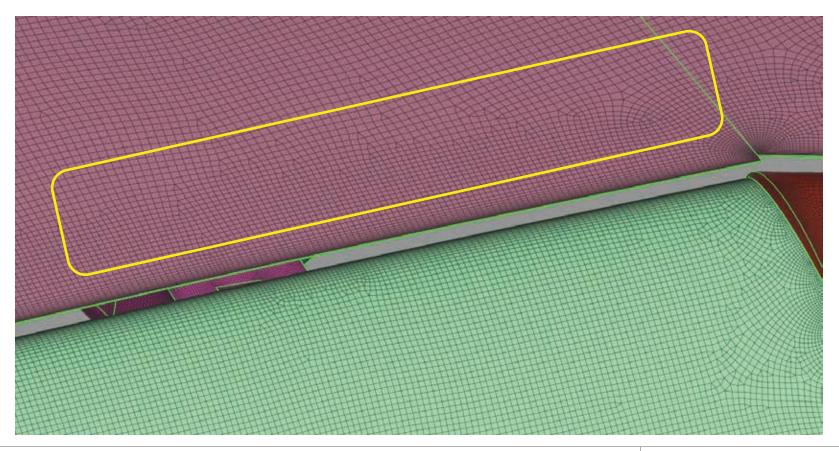
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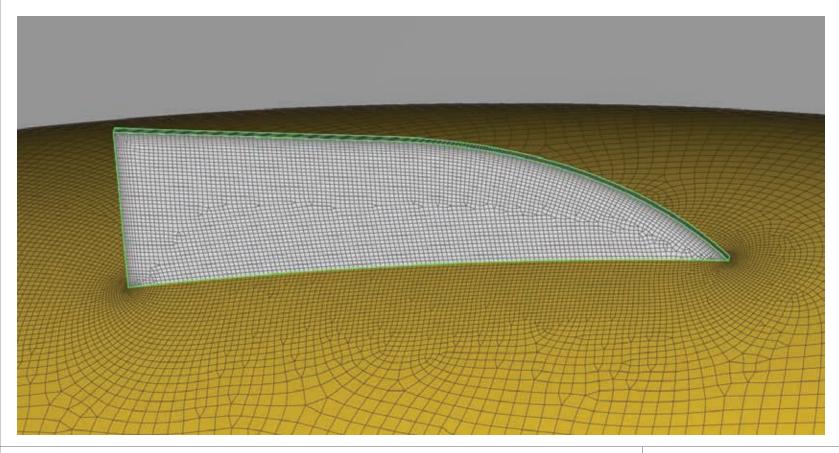
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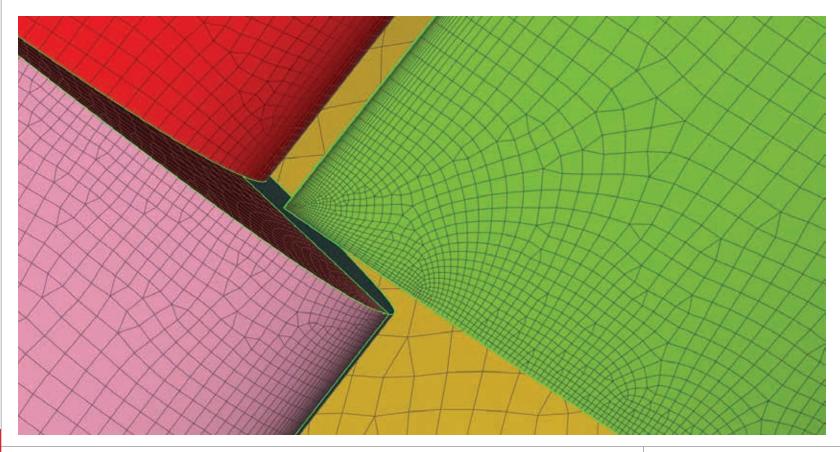
Engine nacelle area



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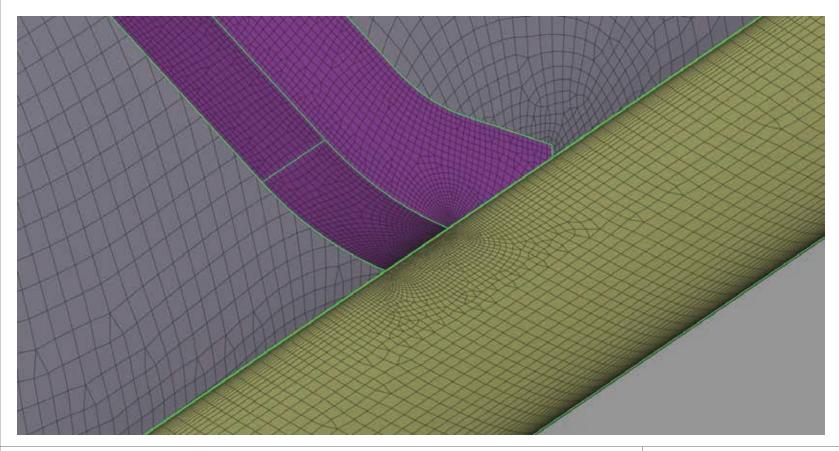
Engine nacelle area



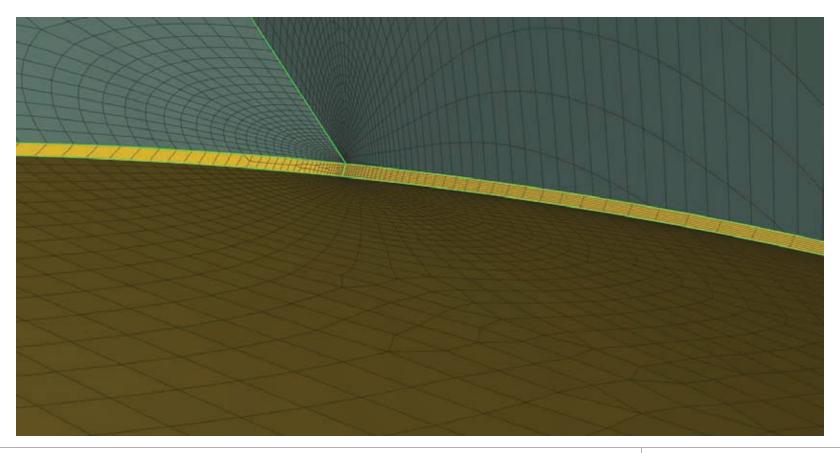
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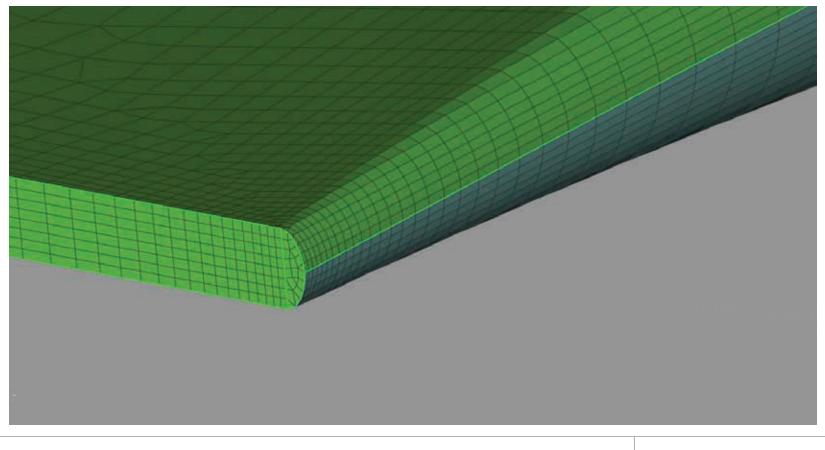
Underside area



Underside area



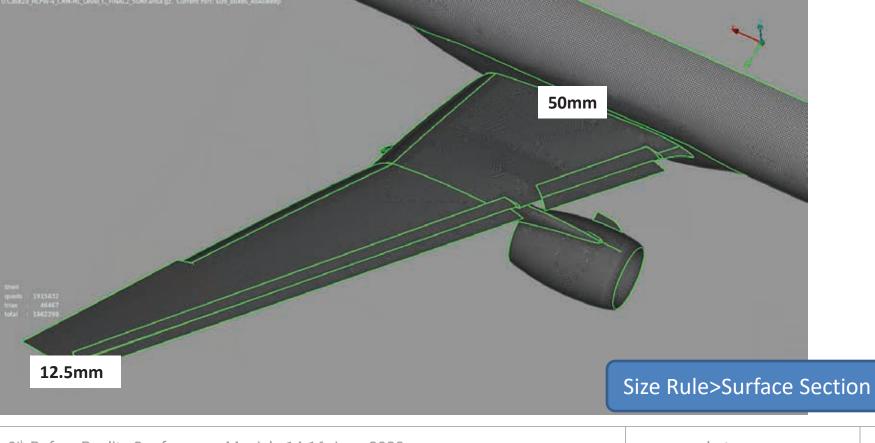
Wingtip area



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Main wing – Linear Variation of maximum length



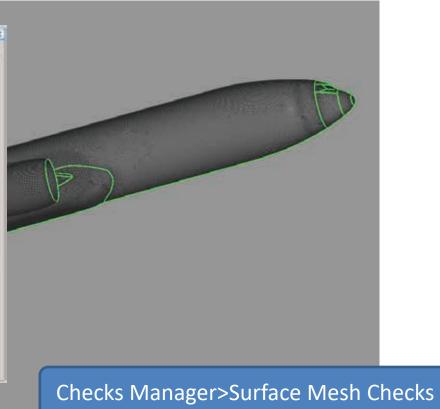
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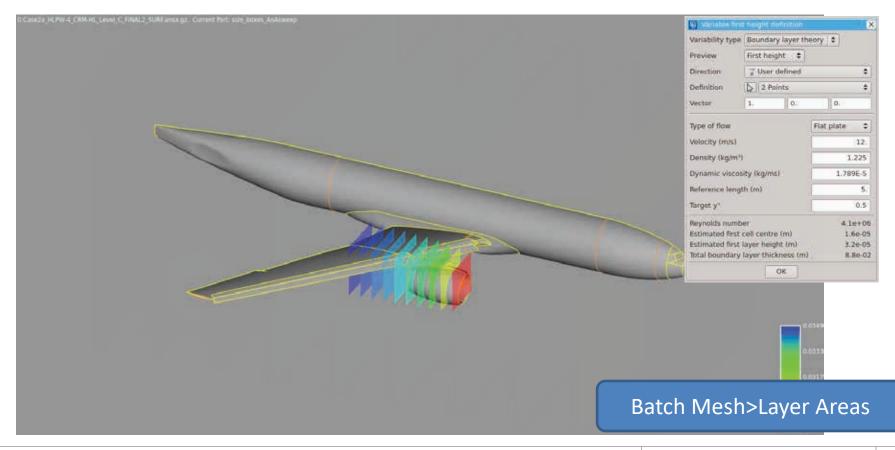
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Surface Mesh Checks

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Variable First Layer height specification

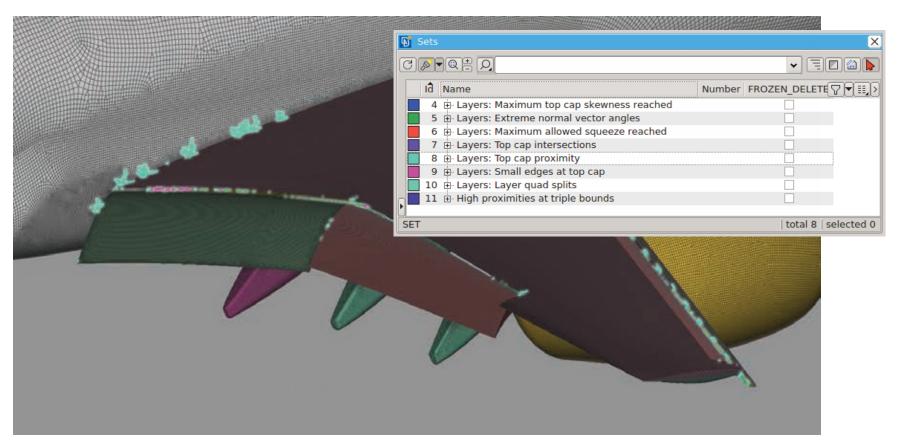


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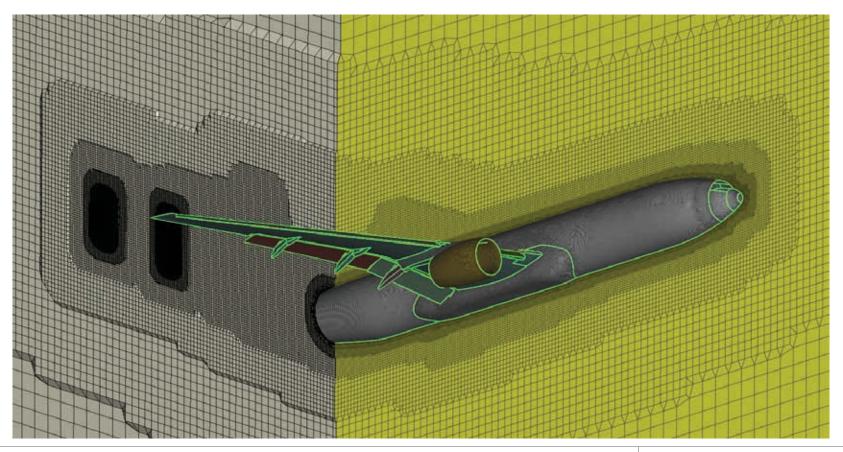
Layers generation – 70 layers variable first height and growth rate



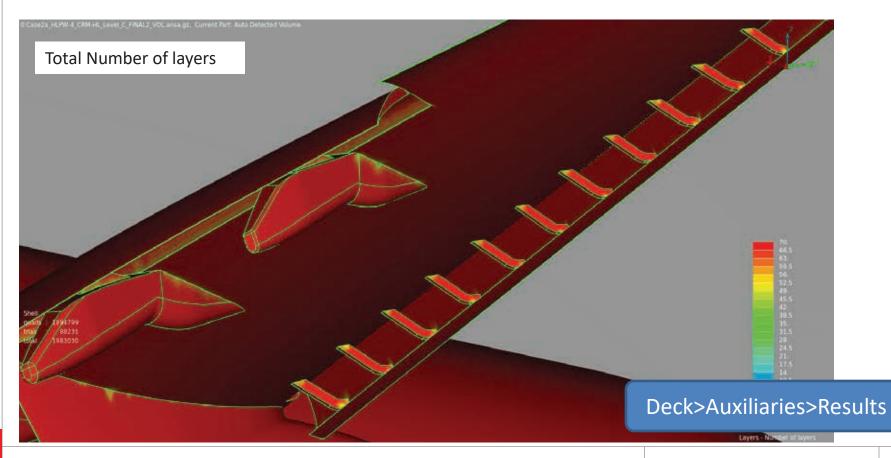
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Volume mesh Level C refinement - 211 million cells (188 million nodes)

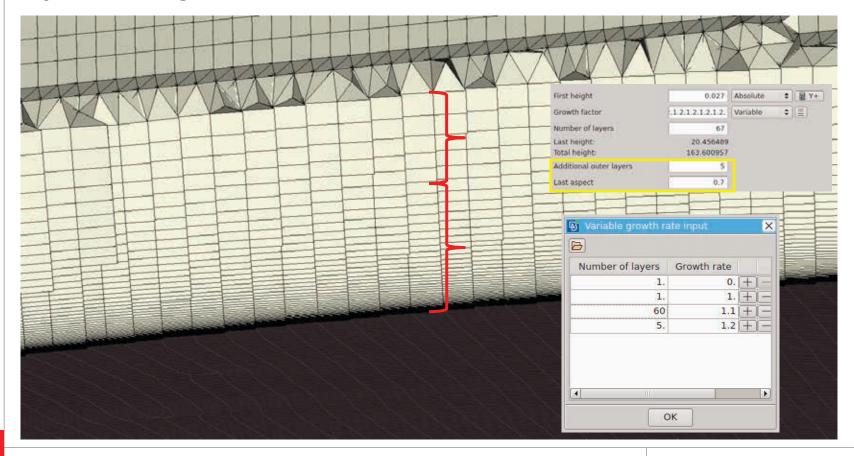


Boundary Layer coverage display



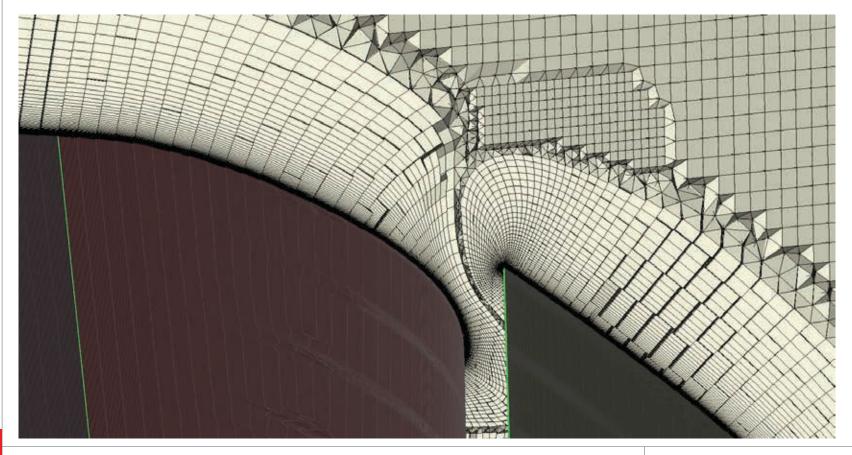
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Layers on wing section



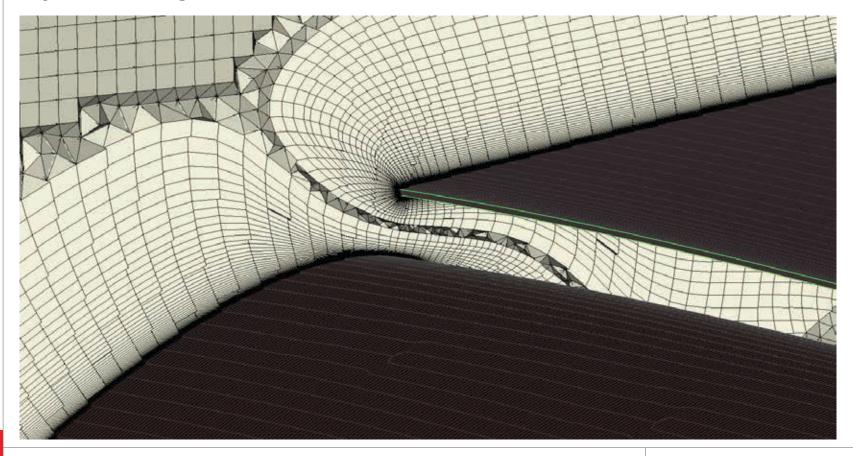
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Layers on wing section



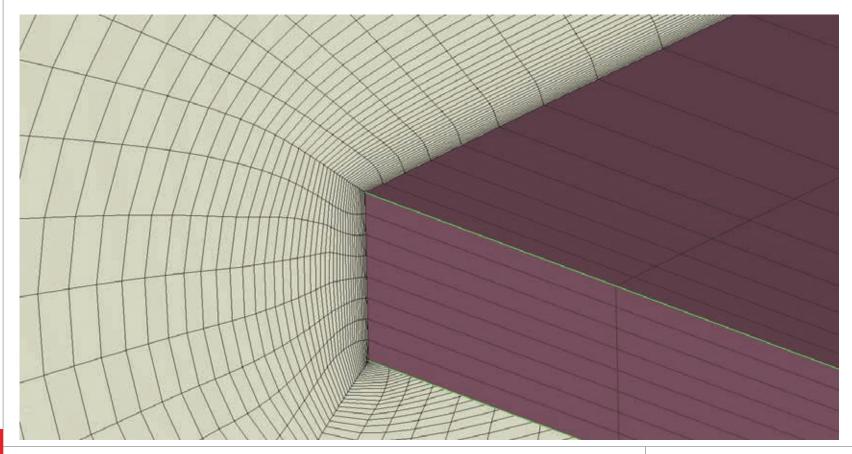
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Layers on wing section

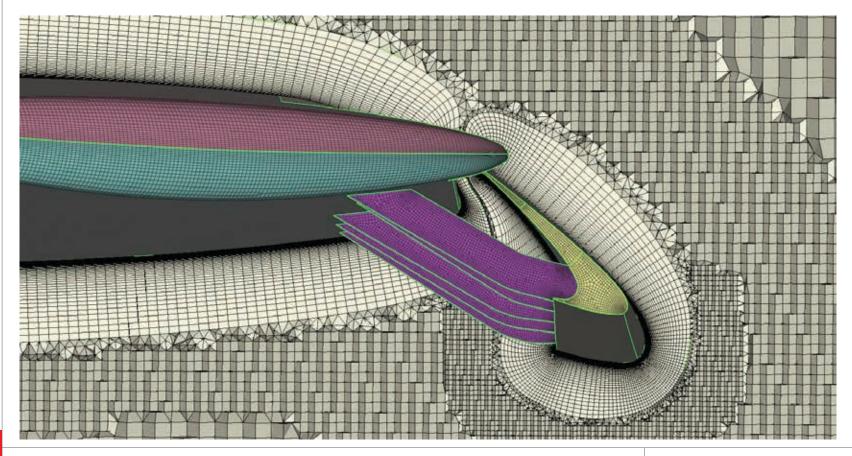


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Orthogonality near the wall

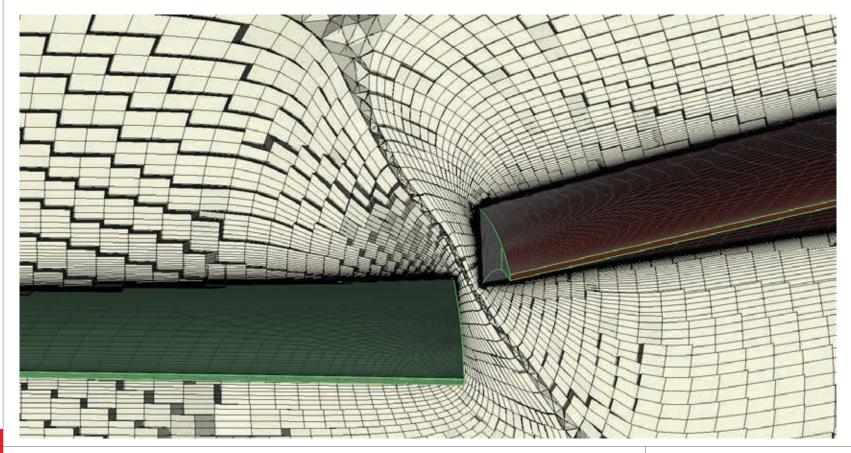


Slat area

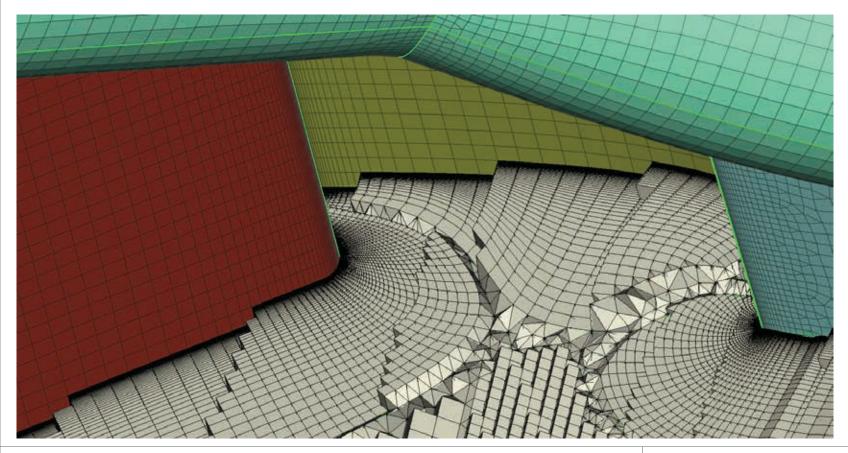


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Gap between the two flaps



Flap underside area



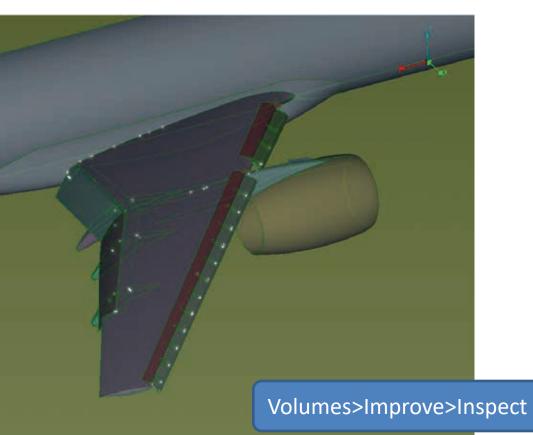
Volume mesh Checks

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* *	Mesh: Free solid faces		Visible Visible	;	0	0 4	0	
	Mesh: Hanging edges Mesh: Unclosed Volume Elements Volumes:Void Areas		Visible Visible Visible					
i i	JP Penetration:Interior Intersections Symmetry Plane		Visible Visible	•	0	0	0	
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Exec	ute							0%
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	Parameters List	ameters List				Check	Summary	
•				•				
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Inspecting mesh quality violations

D Case2a_HLPW-4_CRM-HL_Level_C_FINAL2_VOL.ansa.gz, Current Part: Auto Detected Volum

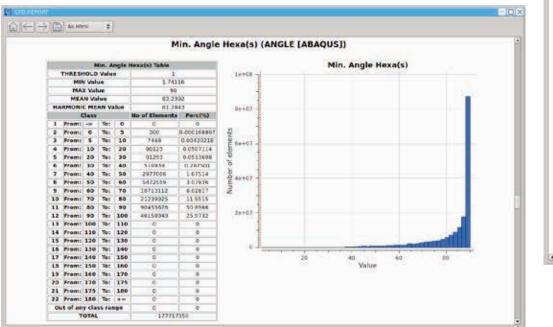
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		e 1
Show	Show Only Hide Mo	we Nodes Fix Quality Recalculate Delete
Course and the second	en e	total 9748 selected 974
Info Pl	et i	
skewness warping min angle max angle max angle max angle	748 off solids :: 2686 off solids (wors : 474 off solids (worst: tetras :: 526 off solids e tetras :: 26 off solids pentas :: 2 off solids e pentas :: 417 off solids	179.39004) (worst: 1.9091E-6) (worst: 179.99918) worst: 0.0622254)

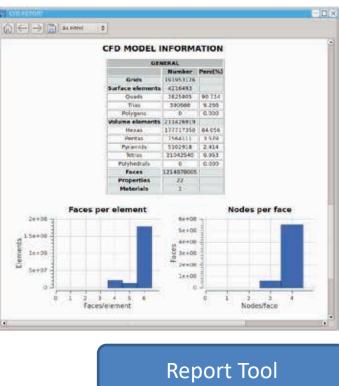


Quality Metrics Reporting

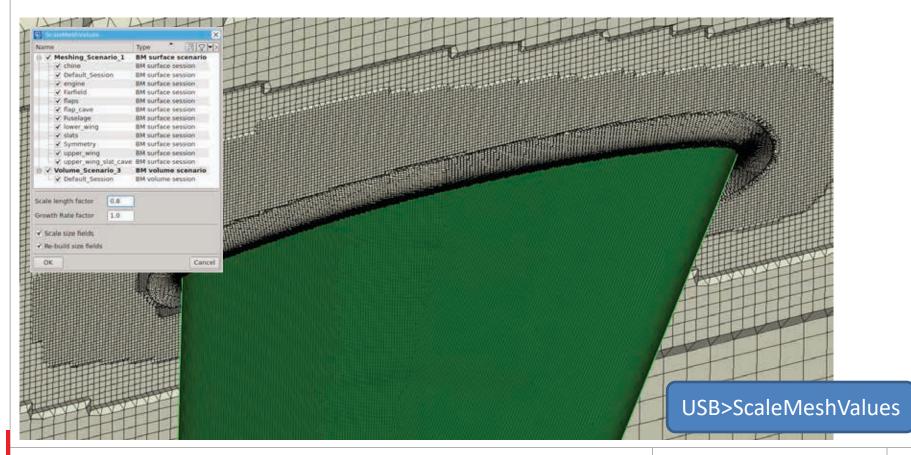
Html creation of mesh statistics

- Mesh quality criteria histograms
- Nodes, Elements, Facets number
- Element type percentages





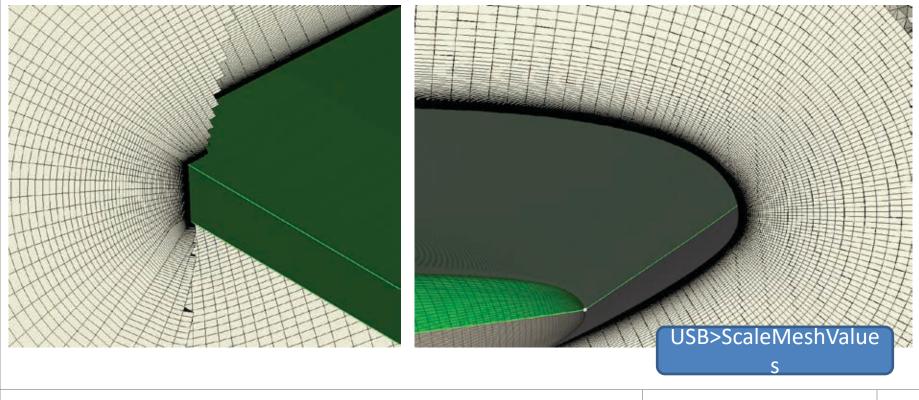
Mesh Refinement Studies on Case 1 of HLPW-5 - five mesh levels



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Mesh Refinement Studies on Case 1 of HLPW-5 - five mesh levels

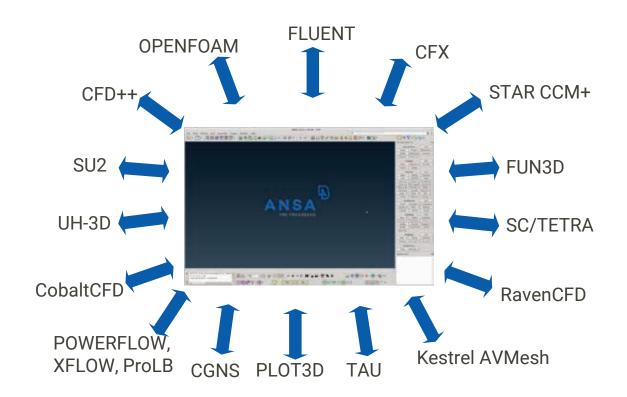
Functionality to automatically scale all length values of Batch Mesh



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Mesh output in multiple CFD mesh formats





Performance - Memory

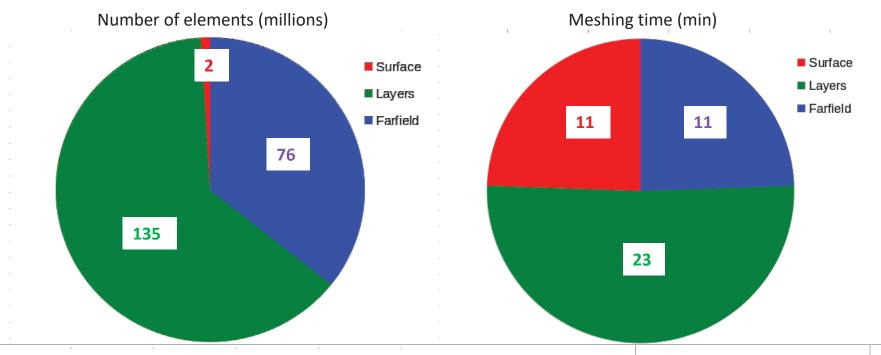
211 million cells Peak memory usage 92 Gb RAM

0.44 Gb/million cells

GENERAL									
	Number	Perc(%)							
Grids	190019251								
Surface elements	2137201								
Quads	2034023	95.172							
Trias	103178	4.828							
Polygons	0	0.000							
Volume elements	211426919								
Hexas	177717350	84.056							
Pentas	7564111	3.578							
Pyramids	5102918	2.414							
Tetras	21042540	9.953							
Polyhedrals	0	0.000							
Faces	1214878005								
Properties	21								
Materials	1								

Performance – Speed

2 million shell elements in 11 min 211 million volume elements in 34 min Surface mesh = 11 million shells/hour Volume mesh = 372 million cells/hour

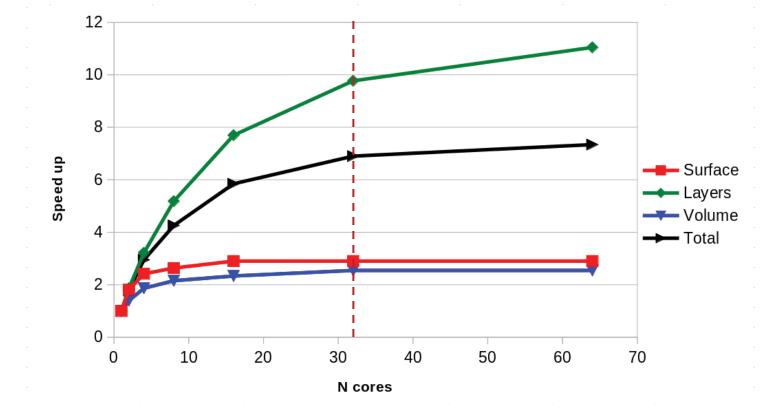


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Speed performance scalability

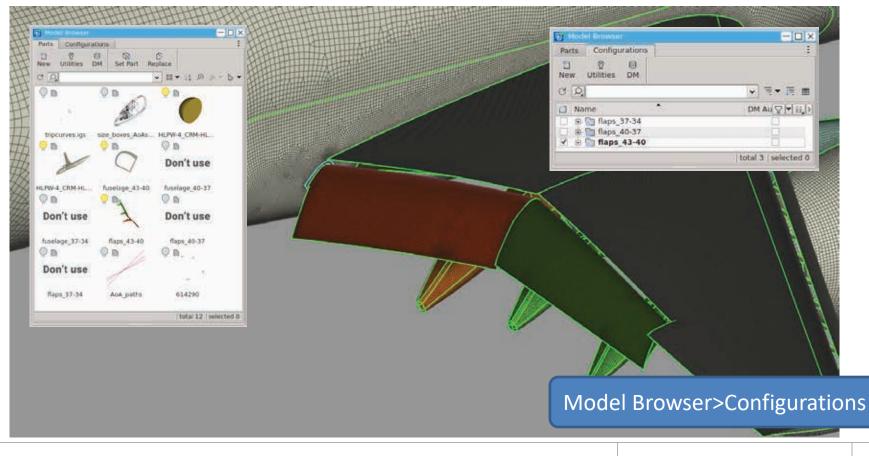


Hardware used has 32 cores (64 threads)

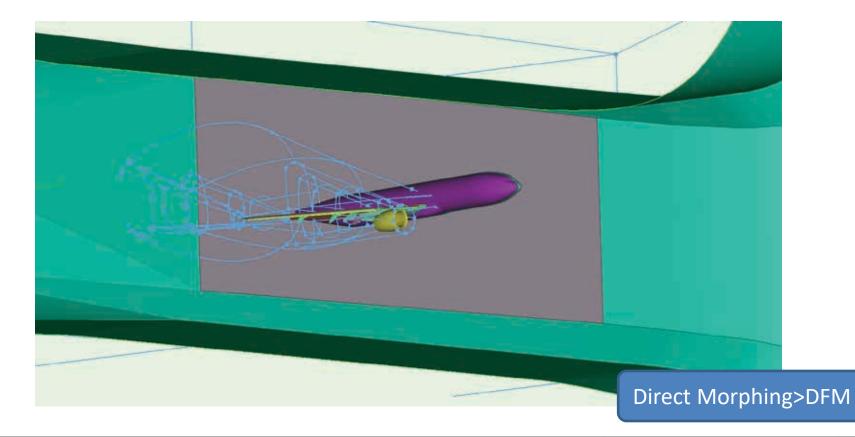
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ß

Use of Configurations for modifying flap angles – HLPW-4

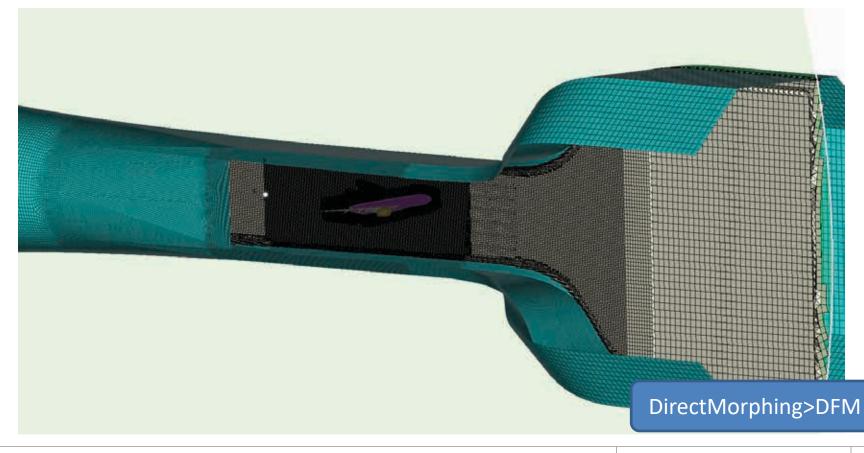


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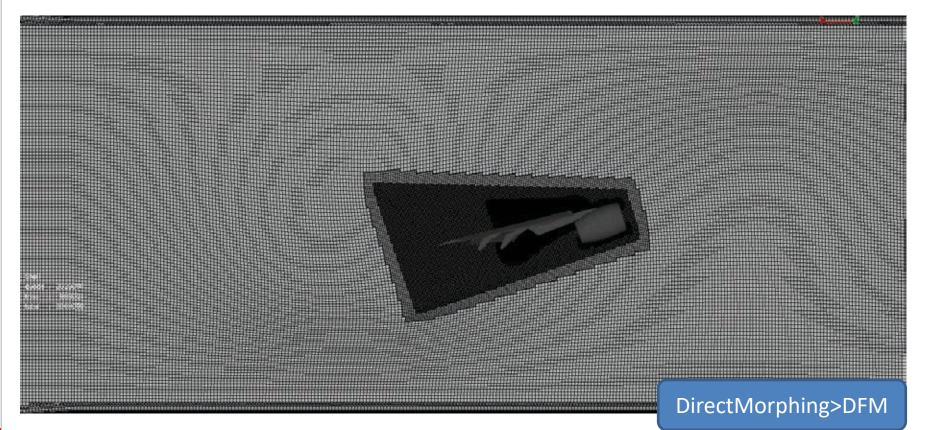


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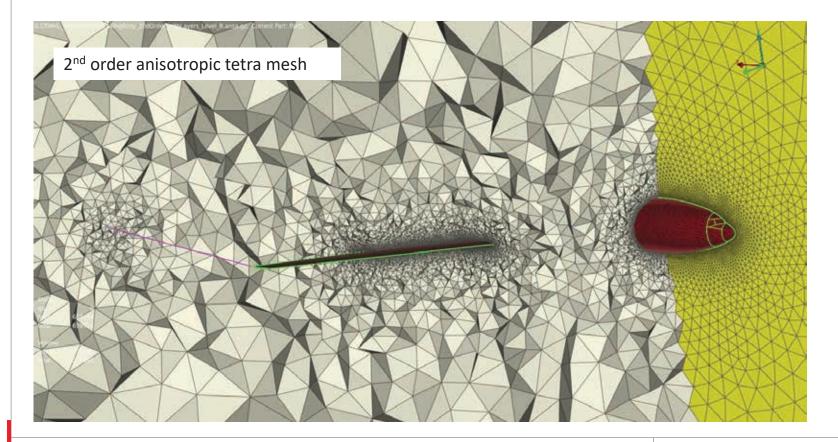
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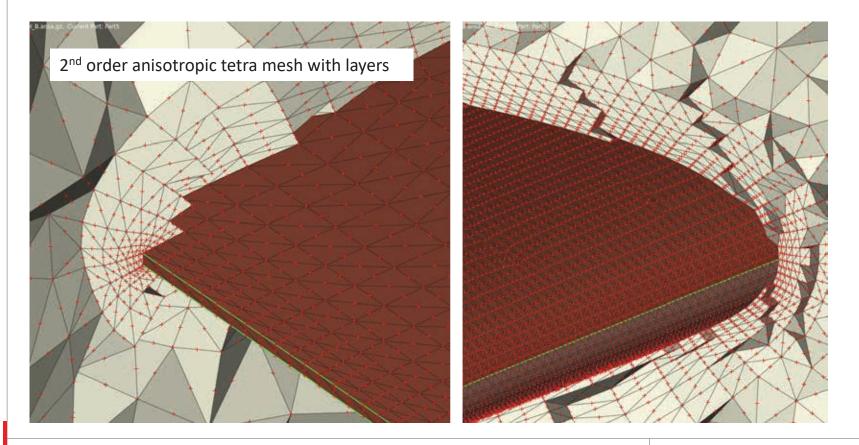
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Meshing for CFD High Order methods – HLPW-5 Case 1 - WMLES methods

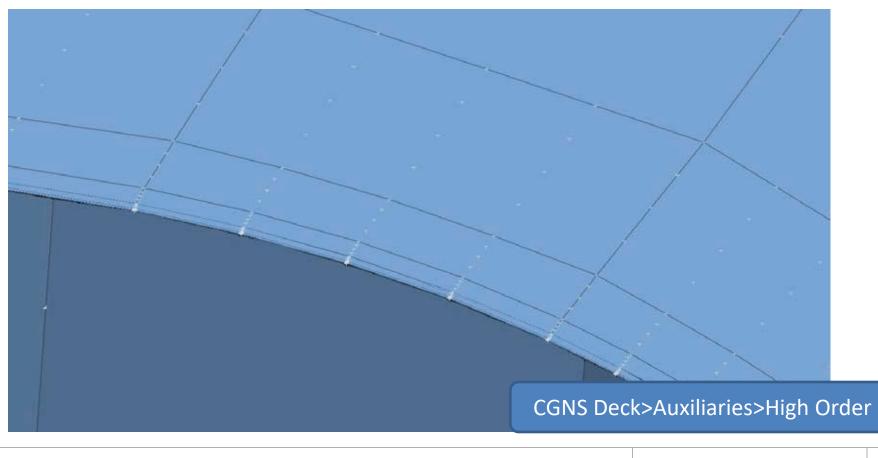




Meshing for CFD High Order methods – HLPW-5 Case 1 - WMLES methods



Mesh elevation to 3rd and 4th order





Recap



Closing remarks

ANSA can generate fully automatically high quality meshes for aerospace models, following precise user controls and specifications.

> ANSA presence is now widespread in the aerospace field (Industry, Academia, AIAA workshops..)

We have achieved this by listening to our customers, working together and learning from them, in order to provide them with the optimum solutions for their work.

We have done this since the beginning of BETA CAE and we have enjoyed every single moment of this journey.





